# Risk Allocation for Design Optimization with Unidentified Statistical Distributions

SciTech AIAA Non-Deterministic Approaches

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- The RBDO community often assumes you can identify statistical distributions
- $\cdot\,$  It is difficult to identify statistical distributions in practice
- Regulators (e.g. FAA) tell you what to do when you can not identify the statistical distribution

Perhaps the regulators are more statistically savvy!

# Obtaining conservative failure allowables

- You have performed a handful of tests on a material
- What failure strength do you use? (failure allowable)
- · Deal with epistemic and aleatory uncertainty
- How to conservatively estimate the failure strength
- Various tolerance interval methods

- 1. Conservative estimation of failure strength
- 2. Non-parametric and Hanson Koopmans tolerance intervals
- 3. Simple risk allocation RBDO for UAV redesign

## Conservative estimate of failure strength



Figure 1: Histogram of 18 tension tests on composite material.

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# One-sided tolerance interval to estimate allowable strength



Figure 2: Estimating the 10th percentile to 95% confidence.

# One-sided tolerance intervals for Aerospace

- 1st percentile to 95% confidence (A-basis): used in non-redundant structures
- 10th percentile to 95% confidence (B-basis): used in redundant structures
- FAA regulations on how to calculate failure strength allowables
- If distribution is known, easy to calculate!

#### It's difficult in practice to identify a known distribution

- You may have too much data
  - Tiny deviations from a distribution are enough to reject that samples come from that distribution
- You may have too little data

I'm not sure where this sweet spot exists...

Consider this sample of 10 (from standard Normal distribution) x = [0.98, -0.7, 0.95, -1.67, -1.4, 0.73, -0.2, 1.76, 1.18, 1.62]

 Table 1: KS-Test 95% confidence: reject distribution if P-value < 0.05</th>

| Distribution | P-value    | Reject? |
|--------------|------------|---------|
| Normal       | 0.57       | False   |
| Lognormal    | 0.57 False |         |
| Weibull      | 0.92       | False   |
| Gamma        | 0.52       | False   |
| Student's t  | 0.57       | False   |

The random sample could have come from just about any distribution!

Order a random sample **x** as

$$x_1 \le x_2 \le \dots \le x_n \tag{1}$$

then the non-parametric tolerance interval for  ${\it P}$  and  $\gamma$  confidence is expressed as

(2)

where *i* is determined from *P* and  $\gamma$ 

Solves the tolerance interval problem with large samples! However, it doesn't work well with small data.

# Non-parametric failure strength on previous 18 samples



**Figure 3:** With just 18 samples, the non-parametric tolerance interval is limited to certain percentile confidence levels.

The tolerance interval is defined as

$$x_j - b(x_j - x_i) \tag{3}$$

where b is solved depending on  $i, j, P, \gamma, N$ 

- *b* has been difficult to solve for (until today...)
- Sporadic use in FAA, SAE, mil-spec...
- Hanson-Koopmans assumes the true distribution is in the Log-Concave CDF class
- Non-parametric assumes the true distribution is continuous

# Hanson-Koopmans failure strength of previous 18 samples



**Figure 4:** With just 18 samples, the Hanson-Koopmans tolerance interval can be calculated for any percentile and confidence.

# My tiny Python library for tolerance intervals

# https: //github.com/cjekel/tolerance\_interval\_py

- Calculate one-sided tolerance intervals for Normal, Lognormal, Non-parametric, and Hanson-Koopmans methods
- Calculate Hanson-Koopmans b for any  $j, N, P, \gamma$

```
import numpy as np
import toleranceinterval as ti
x = np.random.random(100)  # random sample of n=100
# estimate the 10th percentile to 95% confidence
bound = ti.oneside.hanson_koopmans(x, 0.1, 0.95)
```

```
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```

- Fairly common statistical distribution class
- Includes all: Normal, Exponential, Gumbel, Laplace, Logistic, Rayleigh, Maxwell, Uniform, Lognormal, and Pareto distributions
- Also includes subsets of other distributions
- Composite material handbook: composite failure strength generally follows the Log-Concave CDF class

## Visual example of what is Log-Concave CDF



**Figure 5:** The log of the CDF for the Uniform and Normal distributions is concave, while Student's t-distribution in convex.

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# **Risk allocation**

Individual components have different probabilities of failure.

Some potential advantages of risk allocation:

- · Lower weight for same system probability of failure
- Lower system probability of failure for same weight

- Apply Hanson Koopmans methods to do redesign a UAV
- Initial UAV design assumed components have equal safety margins (similar to FAA regulations)
- Redesign the wing and horizontal tail to have difference probabilities of failure

# Initial design of UAV

- Takeoff weight 15 lbs
- Wing weight 1.35 lbs
- Horizontal tail weight 0.3 lbs
- Wingspan 9 ft
- Both wing and tail are fully stressed
- Stress allowable 1st percentile to 95% confidence from Hanson Koopmans on the 18 tests



**Figure 6:** Image of Puma 3 AE by AeroVironment<sup>®</sup> inspired these design specs.

The wing and tail failures are assumed to be independent, thus the system probability of failure is

$$P_f = 1 - (1 - P_w)(1 - P_t)$$
(4)

where  $P_w$  and  $P_t$  are the failure allowables.

The component weight can be assumed to be inversely proportional to change of failure allowable

$$\frac{\sigma_i}{\sigma_n} = \frac{W_n}{W_i} \tag{5}$$

from changing the skin thickness.

## Simple risk allocation RBDO

- Minimize the weight of the wing and horizontal tail
- By changing the allowable failure strength for each component
- Such that the system has the same probability of failure

$$\min w = w_w(p_w, \gamma_w) + w_t(p_t, \gamma_t)$$
(6)

such that: 
$$P_f \le 0.02$$
 (7)

$$\gamma_{\rm W} = \gamma_t = 0.95 \tag{8}$$

#### Results when using the Hanson-Koopmans method



Figure 7: Contour plot of the objective function when using the Hanson-Koopmans method. Charles Jekel · https://jekel.me

## Results along the constrain boundary



**Figure 8:** Line plot of the objective function along the constrain boundary when using the Hanson-Koopmans method.

**Table 2:** Comparison of the UAV specifications from the previousequal safety margin design and the new RBDO optimal design.

|                     | Equal safety margin | RBDO            | RBDO   |
|---------------------|---------------------|-----------------|--------|
|                     | Hanson-Koopmans     | Hanson-Koopmans | Normal |
| $P_f$               | 0.020               | 0.020           | 0.020  |
| $P_W$               | 0.010               | 0.016           | 0.017  |
| $P_t$               | 0.010               | 0.004           | 0.003  |
| w <sub>w</sub> , lb | 1.35                | 1.30            | 1.16   |
| w <sub>t</sub> , lb | 0.30                | 0.32            | 0.27   |
| w, lb               | 1.65                | 1.62            | 1.43   |

- Lighter UAV for the same probability of failure using risk allocation
- Risk allocation between Hanson Koopmans and Normal distribution resulted in similar component failure probabilities
- Hanson Koopmans UAV was  $\approx$  15% heavier than using a Normal distribution

- RBDOs often assume that it is possible to identify statistical distributions
- It is difficult to identify statistical distributions
- Regulation (e.g. FAA) have methods when it is not possible to identify the distribution
- Classes of distributions may be an approach to make RBDOs more robust